INSTRUCTIONS for CONTINUED AIRWORTHINESS

Including INSTALLATION, MAINTENANCE & SERVICE INSTRUCTIONS



Airglas® Model L2700 & L8500 Ski Kits *for* Skid Gear Equipped Helicopters

MANUAL REVISION: C Page Revision 0

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MANUAL REVISION: C Page Revision 0

Airglas MANUAL # L2700/L8500-105

REVISION	ISSUE DATE	Page &	BY	EXPLANATION OF REVISION
NUMBER		Revision #		
Original	31 OCT 2007	-	Christopher Donnelly	Original Document
Α	8 AUG 2008	3,4,6 & 7	Christopher Donnelly	Changed callout for Bell 407 ski kit part numbers
		#1		
В	6 NOV 2008	3#2, 7#2,	Christopher Donnelly	Added model specific instructions, added a Special Note,
		13#1,15#1,		added a Special Note
		16#1		
С	25 MAY 2010	13-16#2	Christopher Donnelly	Added (+/-10%) to inspection interval. Changed 120 day
				inspection interval to 300 hours. Updated contact info.

RECORD OF REVISIONS

Distribution of Changes

In the event that any appreciable changes are made to any portion of this manual number L2700-L8500-105; a new copy of the revised manual or affected pages will be sent to the STC holder that is on record.

MANUAL REVISION: C Page Revision 3

LIST OF EFFECTIVE PAGES

LIST OF REVISIONS

Revision 0 (Original Issue) 31 October 2007

Revised Portions	Pages	Revision No.
COVER	1	0
2 (Blank)	2 (Blank)	0
RECORD OF REVISION	3	3
LIST OF EFFECTIVE PAGES	4	3
DISCRIPTION	5	0
MODEL LIST	6	1
SKI SPECIFICATIONS with WEIGHT and BALANCE	7	2
DEFINITIONS	8	0
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INSTRUCTIONS FOR CONTINUED AIRWORTHINESS	13-16	2
SPECIAL INSTRUCTIONS	17	0
ATTACHMENT 1 PART 29 REQUIREMENTS	18-22	0

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Description

Airglas, Inc. has designed and manufactured for more than forty years, a full length composite ski that dramatically increases the load distribution of the helicopter skids on snow or soft surfaces. Not only does it allow the aircraft to land on snow with full support of the aircraft; it also allows for "running" landings and takeoffs.

The skis add a marginal weight penalty and have no adverse affects on the flight characteristics of the aircraft. The center of gravity changes are negligible. Through years of flight time; no indications of negative effect on flight have been reported. Vibration analysis has typically shown the skis to work harmoniously with the skids.

The skis are attached on virtually all models in the same way. A series of stainless steel straps that wrap around the skid tube secure the skis. Typical installation takes 1-2 hours depending on the crew, environment, and size of aircraft. There is a cutout area that provides clearance for the ground handling wheels to be installed with no interference from the skis.

Since the skis are very simple and have no moving parts; they are very easy to maintain and inspect. This manual will detail the installation, inspection procedures and intervals along with maintenance and repair instructions.

Model Listing of Helicopters and Skis covered by this manual

Aircraft Model	Ski Model
Bell 47J,47J-2,42J-2A	L2700-47J
Bell 204B, UH-1B	L7000 & L8500B
Bell 205A and 205A-1	L8500 & L8500A
Bell 206A and 206B, B3	L2700-206A
Bell 206L and 407	L2700-206L
Bell 206L-1, 206L-3, 206L-4	L2700-206LHS & LHSR
Bell 212	L8500A & R
Bell 412, 412CF, 412EP	L8500A, R & RR
Bell 427	L2700-427
Bell 430	L8500AUT & AUTR
Eurocopter France (Aerospatiale) AS350B, AS350B1, AS350B2, AS350B3, AS350BA, AS350D	L2700-AS350
Eurocopter Canada, Ltd. Bölkow BO-105 LS A-3	L8500-BO105
Fairchild Hiller FH1100	L2700-FH1100
Societe Nationale Industrielle Aerospatiale AS355E & AS355F	L2700-AS350
Aerospatiale SA341G "Gazelle"	L2700-SA341 & HS
Sud Aviation SA318C – Alouette Astazou	L2700-SA318
Hughes 369A, 369H, 369HM, 369HS, 369HE	L2700-369HS
MD Helicopter, Inc. (Hughes) 369D, 369E, 369F, 69FF	L2700-369D & HD

SKID-TYPE SKI SPECIFICATIONS

SEE THE 'MODEL SPECIFIC' INSTALLATION DRAWINGS FOR SPECIAL INSTALLATION & MAINTENANCE INSTRUCTIONS AND WEIGHT & BALANCE DATA

AIRGLAS P/N: L8500A	
Dimensions	163.5″ x 22″
Approximate Area per ski	
Approximate Weight per ski	72 lbs.
C.G. of ski	47.25" Forward of Socket Face
AIRGLAS P/N: L8500R	
Dimensions	
Approximate Area per ski	
Approximate Weight per ski	RH=70 lbs., LH=72 lbs. RH=45.25" LH=47.25" Forward of Socket Face
C.G. OI SKIS	RH=45.25" LH=47.25" FOIWard OF SOCKET FACE
AIRGLAS P/N: L8500RR	
Dimensions	163 5″ x 22″
Approximate Area per ski	
Approximate Weight per ski	
C.G. of ski	
AIRGLAS P/N: L8500AUT	
Dimensions	
Approximate Area per ski	
Approximate Weight per ski	65 lbs.
C.G. of ski	10./" Aft of Fwd. edge of Wheel Well
AIRGLAS P/N: L8500AUTR	
Dimensions	160" x 22"
Approximate Area per ski	
Approximate Weight per ski	
C.G. of ski	
AIRGLAS P/N: L2700-206A	
Dimensions	
Approximate Area per ski	
Approximate Weight per ski	
C.G. of ski	4.5" Forward of Fwd edge of Wheel Well
AIRGLAS P/N: L2700-206LHS	
Dimensions	157.5" x 15.5"
Dimensions Approximate Area per ski	1460 in ²
Dimensions Approximate Area per ski Approximate Weight per ski	1460 in² 40 lbs.
Dimensions Approximate Area per ski	1460 in² 40 lbs.
Dimensions Approximate Area per ski Approximate Weight per ski C.G. of ski AIRGLAS P/N L2700-206LHSR	1460 in ² 40 lbs. 53.75° forward of Socket Face
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5"
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ²
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs.
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs.
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs.
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5"
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 960 in ²
Dimensions	1460 in ² 6 in ² 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 960 in ² 93 lbs.
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 960 in ² 93 lbs.
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Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 118.5" x 13.5" 960 in ² 33 lbs. 920 in ² 923 in ² 923 in ² 25 lbs.
Dimensions	1460 in ² 6 in ² 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 36 lin ² 33 lbs. 36.0" Forward of Socket Face
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 118.5" x 13.5" 960 in ² 33 lbs. 920 in ² 923 in ² 923 in ² 25 lbs.
Dimensions	1460 in ² 40 lbs. 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 118.5" x 13.5" 960 in ² 33 lbs. 31 lbs.
Dimensions	1460 in ² 63.75" forward of Socket Face 53.75" forward of Socket Face 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 960 in ² 33 lbs. 36.0" Forward of Socket Face 116" x 14" 25 lbs. 135" Forward of Fwd. edge of Wheel Well
Dimensions	1460 in ² 6 in ² 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 360 in ² 31 lbs. 36.0" Forward of Socket Face 116" x 14" 25 lbs. 138.0" x 13.7" 138.0" x 13.7"
Dimensions	
Dimensions	
Dimensions	
Dimensions	1460 in ² 53.75" forward of Socket Face 157.5" x 15.5" 1460 in ² 38 lbs. 54.25" forward of Socket Face 118.5" x 13.5" 960 in ² 31 lbs. 36.0" Forward of Socket Face 116" x 14" 25 lbs. 138.0" x 13.7" 138.0" x 13.7" 135 in ² 30 lbs. 41.5" Forward of Socket Face
Dimensions	
Dimensions	
Dimensions	

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Definitions

- FAA/Authority = Federal Aviation Administration or another airworthiness authority
- FAR = Federal Aviation Regulation
- ICA = Instructions for Continued Airworthiness
- JAR = Joint Airworthiness Regulations
- LOAP = List of Applicable Publications
- TDC = Type Design Changes
- STC = Supplemental Type Certificate

THIS MANUAL INCLUDES INFORMATION PROPRIETARY TO AIRGLAS, INC. AND SHALL NOT BE USED TO MANUFACTURE OR REPRODUCE ANY PART OR ASSEMBLY WITHOUT THE PRIOR WRITTEN PERMISSION OF *Airglas, Inc.*

INITIAL INSTALLATION INSTRUCTIONS

(See Installation Drawings attached for additional detail)

1) Removal of skis from packaging:

- a) Remove outer burlap wrap by cutting/removing string (if used).
- b) Remove shrink-wrap and foam from around skis. Use care.(Do not to cut through the foam.)
- c) Remove documents & drawings for use in installing the ski set.



Packaged ski as shipped without burlap.



Ski with 2x3 stiffener still attached after packaging has been removed.

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Continued

Remove nuts (AN365-524 or MS21042-5), washers (AN960-516) from each mounting screw using care not to damage parts; they will be used to install ski on helicopter.

 d) Remove straps from ski and inventory straps for use in final installation.
 Note: Front Strap (Typical) is double stacked and located with the strap in front of the cutout. The Double Strap (Typical) can be removed by hand.

2) Attaching skis on the skid tubes:

- a) Jack up the helicopter using the Factory approved jacking points or an approved alternative jacking system.
- b) Remove all skid plates and plug holes with flush set screws or a removable sealant to prevent the introduction of water or contaminants to skid tube.

For the alternative lifting procedure of using the <u>Ground Handling Wheels</u>; complete steps c. thru f.

- c) Raise the helicopter using the ground handling wheels.
- d) Position ski behind the skid in preparation to slide ski between ground handling wheels.
- e) Lift tail of ski up to the point that the tip of the ski will slide under the skid.

f) Slide ski between tires with a forward motion. The ski will have to be twisted in order to miss the tires as the front shovel section passes the wheels.

g) Lift each ski into place on the skid tube, making sure to have adequate personnel to hold the ski in place while other personnel place the straps across the skid tube.

h) Place straps in the appropriate locations across skid tube and on to the ski attach studs. (SEE Installation Drawing)

i) Attach and <u>Hand Tighten Only</u> the hardware on the studs.



Remove Skid Shoes from skis.

Position ski to insure the skid is tight against the socket face.

3) Tighten and torque attaching hardware:

- a) Lower helicopter to the ground in order to position and seat the skis.
- b) Make sure the straps are positioned at equal thread depths on the studs from inboard to outboard.
- c) Torque nuts to 36 inch pounds (+/- 5). See notes on installation drawing.

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Leave straps loose until full weight is on the ski.



Tension straps evenly on each side of skid tube.

REMOVAL INSTRUCTIONS

- 1) Remove skis from the skid by reversing installation process.
 - a) Remove nuts and washers holding the straps on.
 - b) Remove the straps from the skis.
 - c) Lift or fly the helicopter off of the skis.
 - d) Replace straps and hardware on ski to avoid loss.

Caution: Replace self locking nuts after two installations.

Instructions for Continued Airworthiness WITH Airglas, Inc. Skid Gear Skis

Airworthiness Limitations

(1) Each mandatory replacement time:

a) There are NO mandatory replacement time limited parts on Airglas Ski Kits.

Daily Inspection

(+/- 10%)

A visual inspection is required prior to each flight for overall condition of ski, skid attachment bracket slippage marks, and all associated hardware condition.

- Inspect for loose or stripped strap attach screws or damage to attaching clamps.
 a) Replace any damaged attaching parts before next flight.
- II) Inspect for cracks, holes or abraded areas in the fiberglass.
 - a) If minor (field repairable damage); complete repair within 120 days of noted discrepancy.
 - b) If major (manufacturer required repairs); repair or replace before next flight.
- III) Inspect for loose, cracked or working rivets.a) Replace within 120 days of noted discrepancy.
- IV) Inspect for worn, cracked or missing runners.a) Replace within 120 days of noted discrepancy.

SPECIAL NOTE:

Some installation drawings may contain special instructions or may also require specific hardware for the applicable ski kit.

Be sure to review the installation drawings that are specific to the ski model during installation and maintenance intervals.

Periodic Inspection

A **300 HOUR INSPECTION** (+/-10%) is required and includes removal of ski to thoroughly inspect for damage. This inspection requires:

- Inspect for loose or stripped strap mounting screws.
 a) Replace any damage parts before next flight.
- 2) Inspect for cracks or holes and/or abrasion areas in fiberglass.a) Repair or return to factory for repair before next flight.
- 3) Inspect for loose, cracked or working rivets.a) Replace before next flight.
- 4) Inspect for missing or cracked runners.a) Replace before next flight.
- 5) Inspect for corrosion on the bottom of skid tube and at the heel cap.a) Refer to Helicopter Factory Maintenance Procedures for any corrosion control issues.

Types of Possible Damage

Negligible Damage

- 1) Small and shallow nicks, scratches and abraded areas on the top or bottom.
- 2) Stress Cracks in the gel coat.

Field Repairable Damage

- 1) Replacement of studs i.e. strap mounting screws.
- 2) Cracks or fractures less than 3"; more than 3" require consultation with Airglas.
- 3) Delamination of 0.5" horizontal penetration from edge and 3-5" in length.
- 4) Small holes that are no more than 1-2 inches in diameter.
- 5) Abrasions to the ski from terrain contact.
- 6) Replacement of worn runners.
- 7) Replacement of loose or missing rivets.

NOTE: Stress Cracks in the gel coat from flexing are <u>cosmetic</u> and <u>ARE NOT</u> an airworthiness issue.

Non-Field Repairable Damage (Factory Repair Only)

- 1) Strap mounting screws pulled through the ski.
- 2) Delamination within 1" of a mounting screw.
- 3) Cracks and delaminations longer than 3"

Replacement of Straps and Strap Mounting Screws

- 1) Remove Nut (AN365-524 or MS21042-5) and washer from screw.
- 2) Remove attaching strap assembly.
- 3) Remove Nut (AN316-5R) and washer from screw.
- 4) Remove old screw (NAS514P524-24P) from ski.
 - i) This may require chipping or grinding the coating material from around the screw head on the ski bottom.
- ii) Lightly tap the damaged screw through the ski with an appropriate hammer.
- 5) Replace the screw and nut. (See installation drawing appropriate for ski model.)

Rivet / Runner Replacement

Special Note:

Plastic Runners that are attached by screws will not require any special tools or procedures for removal and replacement. Standard mechanics practices will apply.

- 1) Place the ski on a solid surface; drill the heads only off of the rivets with #9 drill bit.
- 2) Drive the rivet shanks through the ski with a 3/16 straight punch.
- 3) Remove the damaged runner from the ski.
- 4) Inspect ski for damage around runner. Make repairs to the ski as necessary.
- 5) Position the new runner in the same location of old runner.
- 6) Align the runner holes with an awl, drift punch or #9 drill bit.
- 7) Clamp the runner to the ski with enough clamps to maintain correct positioning.
- 8) Install SSB6-8 stainless rivets (Available from Airglas, Inc.) using an appropriate rivet puller.
- 9) Grind rivet stems flush with the surface of the runner.
- 10) Touch up with flat black paint as necessary.
- 11) Heat ski base & runners to 200°F with a heat gun and apply a coating of paraffin wax.
- 12) Inspect replacement runner installation and return ski to service.

Base Surface Maintenance

If the bottom surface sustains excessive wear; it may be sanded down with a 36 grit belt or disc sander and then recoated with epoxy or abrasion resistant gel coat. Once the epoxy or abrasion resistant gel coat is cured, the surface should be re-sanded with 80 grit paper using an orbiting sander. When more advanced and complicated repairs are necessary; *Airglas, Inc.* should be consulted.

Note 1: Excessive Wear is defined as when the ski base coating is scraped or worn off to the point that the underlying fiberglass composite fibers are exposed.

Note 2: Epoxy and abrasion resistant gel coat formulas change over time. Contact: *Airglas, Inc.* for current information.

THE AIRWORTHINESS LIMITATIONS SECTION IS FAA APPROVED AND SPECIFIES INSPECTIONS AND OTHER MAINTENANCE REQUIRED UNDER §§ 43.16 AND 91.403 OF THE FEDERAL AVIATION REGULATIONS UNLESS AN ALTERNATIVE PROGRAM HAS BEEN FAA APPROVED.

For any other information, or questions, comments or concerns; contact us at:

Airglas, Inc. 3500 O'Malley Road Anchorage, Alaska 99507

907-344-1450 phone 907-349-4938 fax info@airglas.com

SPECIAL INSTRUCTIONS

In case of a hard landing; the skis should be inspected for damages to the ski attaching hardware and to the skis themselves. Inspect for stretched, sheared or loose screws or nuts. Inspect for distortion of the straps. Inspect for cracks in the skis or loosened runners. See pages 13-16 for detailed procedures. The BELL Helicopter Operators Manual Special Instructions section should be reviewed for additional inspection criteria.

In case of a lightning strike; the *Airglas, Inc.* Ski Kits have no specific inspection requirements. See the BELL Helicopter Operators Manual Special Instructions section for lightening strike inspection criteria.

The following is a breakdown of Appendix A to FAR/JAR Part 29 and is intended to provide guidance to assist an applicant for a Type Design Change under a Type Certificate (TC), Supplemental Type Certificate (STC), or Field Approval (FA) requiring Instructions for Continued Airworthiness (ICA). This breakdown is intended to provide guidance to assist an applicant in understanding the ICA requirements of § 29.1529. An applicant may use the guidance to prepare the ICA. Completion of this appendix will provide information needed for the evaluation and will reduce the time required for evaluation of the proposed ICA. The open parentheses () in the Requirement column indicates the status of ICA Requirements; Y = applicable; N/A = non-applicable. In the Location column, list the page number in the applicant's ICA that contains the information.

Requirement	Regulation	Location
(N/A) ICA for each engine.	A29.1(b)	N/A
(N/A) ICA for each rotor.	A29.1(b)	N/A
(Y) ICA for each appliance required by this chapter.	A29.1(b)	All
(Y) Any required information relating to the interface of the	A29.1(b)	
(Y) appliances, (N/A) engines and (N/A) rotors with the rotorcraft.		
(Y) If ICA are not supplied by the manufacturer of an (Y) appliance, (N/A	A29.1	13-17
) engine or (N/A) rotor installed in the rotorcraft, the ICA for the		
rotorcraft must include (Y) the information essential to the continued		
airworthiness of the rotorcraft.		
(Y) A program showing how changes to the applicant's ICA will be	A29.1(c)	3
distributed.		
(N/A) A program showing how changes to the ICA of the manufacture	A29.1(c)	N/A
of the engine(s), rotor(s) and appliances installed in the rotorcraft will		
be distributed, if referenced in applicant's ICA		
(Y) ICA must be in the form of a manual or manuals as appropriate for	A29.2(a)	All
the quantity of data.		
(Y) A format of the manual or manuals which must provide for a	A29.2(b)	All
practical arrangement.	-	
(Y) Content must be prepared in the English language.	A29.3	All
(Y) Introduction information that includes (Y) an explanation of the	A29.3(a)(1)	5-17
rotorcraft's features and (Y) data to the extent necessary for		
maintenance and preventive maintenance.		

FIGURE 1

PARI 29 REQUIREMENTS	Degulation	Location
Requirement	Regulation	Location
(Y) A description of the (N/A) rotorcraft and its systems and	A29.3(a)(2)	5-12
installations, (N/A) engines and its systems and installations,		
(N/A) rotors and its systems and installations, and (Y)		
appliances and its systems and installations.	420 2/2/21	N/A
(N/A) Basic control and operating information describing (N/A) how the rotorcraft components and systems are controlled and (N/A) how the	A29.3(a)(3)	IN/A
rotorcraft components and systems are operated including (N/A) now the		
special procedure and limitations.		
(N/A) Servicing information that covers details regarding (N/A)	A29.3(a)(4)	N/A
servicing points, (N/A) capacities of tanks, (N/A) capacities of	A29.3(d)(4)	N/A
reservoirs, (N/A) types of fluids to be used, and (N/A) pressures		
applicable to the various systems.		
(N/A) Location of access panels for (N/A) inspection and (N/A)	A29.3 (a)(4)	N/A
servicing.	A20.0 (0)(4)	1 1/7
(N/A) Servicing information that covers details regarding (N/A)	A29.3(a)(4)	N/A
locations of lubrication points, and (N/A) the lubricant to be used.	/120/0(4)(4)	1.07.1
(N/A) Equipment required for servicing.	A29.3(a)(4)	N/A
(N/A) Tow instructions and limitations.	A29.3(a)(4)	N/A
(N/A) Mooring information.	A29.3(a)(4)	N/A
(Y) Jacking information.	A29.3(a)(4)	10
(N/A) Leveling information.	A29.3(a)(4)	N/A
(Y) Scheduling information for each part of the (N/A) rotorcraft that	A29.3(b)(1)	N/A
provides the recommended periods at which they should be		
(N/A) cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested,		
(N/A) lubricated and (N/A) the work recommended at these periods.		
(N/A) Scheduling information for the (N/A) rotorcraft's engine(s) that	A29.3(b)(1)	N/A
provides the recommended periods at which they should be (N/A)		
cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested, (N/A)		
lubricated and (N/A) the work recommended at these periods.		
NOTE: This information may be in the FAA/AUTHORITY-accepted engine		
ICA.		
(N/A) Scheduling information for the (N/A) rotorcraft's auxiliary power	A29.3(b)(1)	N/A
unit(s) (APU) that provides the recommended periods they should be		
(N/A) cleaned, (N/A) inspected, (N/A) adjusted,		
(N/A) tested, (N/A) lubricated, and (N/A) the work recommended at		
these periods.		

Figure 1 (continued)

PART 29 REQUIREMENTS		
Requirement	Regulation	Location
(N/A) Scheduling information for the (N/A) rotorcraft's rotor(s) that provides the recommended periods at which they should be (N/A) cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested, (N/A) lubricated, and (N/A) the work recommended at these periods.	A29.3(b)(1)	N/A
(N/A) Scheduling information for the (N/A) rotorcraft's accessories that provides the recommended periods at which they should be (N/A) cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested, (N/A) lubricated, and (N/A) the work recommended at these periods.	A29.3(b)(1)	N/A
(N/A) Scheduling information for the (N/A) rotorcraft's instruments that provides the recommended periods at which they should be (N/A) cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested, (N/A) lubricated, and (N/A) the work recommended at these periods.	A29.3(b)(1)	N/A
(N/A) Scheduling information for the (N/A) rotorcraft's equipment that provides the recommended periods at which they should (N/A) cleaned, (N/A) inspected, (N/A) adjusted, (N/A) tested, (N/A) lubricated, and (N/A) the work recommended at these periods.	A29.3(b)(1)	N/A
(Y) The degree of inspection for each part of the (Y) rotorcraft and its (N/A) engine(s), (N/A) auxiliary power unit, (N/A) rotor(s), (Y) accessories, (N/A) Instruments, and (N/A) equipment.	A29.3(b)(1)	13-14
(N/A) The applicable wear tolerances	A29.3(b)(1)	N/A
The applicant may refer to an (N/A)accessory, (N/A) instrument, or (Y) equipment manufacturer as the source of this information if the applicant shows (Y) that the item has an exceptionally high degree of complexity requiring specialized maintenance techniques, test equipment, or expertise.	A29.3(b)(1)	13-17
(N/A) The recommended overhaul periods and necessary cross references to the Airworthiness Limitation Section.	A29.3(b)(1)	N/A
(Y) An inspection program that includes (Y) the frequency and (Y) extent of the inspection necessary to provide for the continued airworthiness of the rotorcraft.	A29.3(b)(1)	13-14
(N/A) Troubleshooting information describing (N/A) problem malfunctions, (N/A) how to recognize those malfunctions, and (N/A) the remedial action for those malfunctions.	A29.3(b)(2)	N/A
(N/A) Information describing the order and method of (N/A) removing and (N/A) replacing engine(s) with any necessary	A29.3(b)(3)	N/A

Figure 1 (continued)

Requirement	Regulation	Location
(N/A) Information describing the order and method of (N/A) removing and (N/A) replacing rotor(s) with any necessary precautions to be taken.	A29.3(b)(3)	N/A
(Y) Information describing the order and method of (Y) removing and (Y) replacing parts with any necessary precautions to be taken.	A29.3(b)(3)	9-12
(N/A) Other general procedural instructions including (N/A) storage limitations and procedures for (N/A) testing system during ground running, (N/A) making symmetry checks, (N/A) weighing and determining the center of gravity, (N/A) lifting, and (N/A) shoring.	A29.3(b)(4)	N/A
(N/A) Diagrams of structural access plates and information needed to gain access for inspections when access plates are not provided.	A29.3(c)	N/A
(N/A) Details for the application of special inspection techniques including radiographic and ultrasonic testing where such processes are specified.	A29.3(d)	N/A
(N/A) Information needed to apply projective treatment to structure after inspection.	A29.3(e)	N/A
(Y) All data relative to structural fasteners such as (Y) identification, (N/A) discarded recommendations, and (Y) torque values.	A29.3(f)	Installation Drawing
(N/A) A list of special tools needed.	A29.3(g)	N/A
(Y) The Instructions for Continued Airworthiness must contain a section, titled Airworthiness Limitations that is (Y) segregated and (Y) clearly distinguishable from the rest of the document. NOTE: <u>The Airworthiness Limitations Section in the applicant's ICA will be evaluated by the appropriate FAA/AUTHORTY.</u>	A29.4	13-17
(Y) The Airworthiness Limitations Section must set forth each mandatory replacement time, structural inspection procedure approved under § 29.571.	A29.4	13-14
(N/A) If the Instructions for Continued Airworthiness consist of multiple documents, the Airworthiness Limitations Section required by this paragraph must be included in the principal manual.	A29.4	N/A

Figure 1 (continued)

Requirement	Regulation	Location
(Y) The Airworthiness Limitations Section must contain a legible	A29.4	16
statement in a prominent location indicating that the Airworthiness Limitations Section is FAA/AUTHORITY-approved and specifies required maintenance and/or inspections. The exact, required wording of this statement is found in the FAR/JAR.		

Figure 1 (continued)

NOTE: The Airworthiness Limitations Section (ALS) is evaluated and approved by the FAA/AUTHORITY.

~END~

MANUAL REVISION: C Page Revision 0