



European Aviation Safety Agency

SUPPLEMENTAL TYPE CERTIFICATE

10047850

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRGLAS, Inc.

**3500 O'MALLEY ROAD
ANCHORAGE AK 99507
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : EASA.IM.R.121

Type Certificate Holder : ROBINSON HELICOPTER COMPANY

Type Design - Model : R44, R44II

Original STC Number : FAA STC SR02422AK

Description of Design Change:

Fabrication and installation of composite floorboard scuff plates.

Fabrication and installation of Airglas, Inc. LF2600-R44 composite floorboard scuff plates in place of original carpet floor covering.


EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 24 January 2014


Massimo MAZZOLETTI
Certification Manager
Rotorcraft, Balloons, Airships

Note:

The following numbers are listed on the certificate:
EASA current Project Number: 0010028371-001

SUPPLEMENTAL TYPE CERTIFICATE - 10047850 - AIRGLAS, Inc.

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Associated Technical Documentation:

Descriptive Data List LF-2600-R44, revision (-) dated July 10, 2013

Airglas, Inc. Installation, Maintenance, and Instructions for Continued Airworthiness, document number LF2600-R44-105, revision (-), dated July 8, 2013

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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