European Aviation Safety Agency



SUPPLEMENTAL TYPE CERTIFICATE 10050764

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRGLAS, Inc.

3500 O'MALLEY ROAD ANCHORAGE AK 99507 USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.R.121

Type Certificate Holder: ROBINSON HELICOPTER COMPANY

Type Design - Model: R 44, R 44 II
Original STC Number: FAA SR02431AK

Description of Design Change:

Installation of the LAM-R44 Portable Device Mount

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 09 October 2014

│ Massimo MAZZOLETTI Head of Rotorcraft Department

Note:

The following numbers are listed on the certificate: EASA current Project Number; 0010033601-001

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Associated Technical Documentation:

Rotorcraft Flight Manual Supplement AI-LRSCE-R44-FM, Rev. Orig., dated August 20, 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Descriptive Data List (DDL) #LAM-R44, Rev. C, dated August 20, 2014 Instructions for Continued Airworthiness (ICA) LAM-R44-105, Rev. D, dated August 20, 2014 or later revisions of the above listed documents accepted by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

Note:

The following numbers are listed on the certificate: EASA current Project Number: 0010033601-001

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