

SUPPLEMENTAL TYPE CERTIFICATE

10053763

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

AIRGLAS, Inc.

**3500 O'MALLEY ROAD
ANCHORAGE AK 99507
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.042
Type Certificate Holder: GA8 AIRVAN PTY. LTD.
Type: GA-8
Model: GA-8, GA-8-TC 320
Original STC Number: FAA STC SA02441AK

Description of Design Change:

Installation of 8.50" x 6", 6-ply nose tires utilizing an Airglas, Inc. part number PA32-206LA-GA8 Nose Fork Kit.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 22 June 2015



Yves MORIER

**Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)**

10036311
SUPPLEMENTAL TYPE CERTIFICATE - 10053763 - AIRGLAS, Inc. - 300481



Associated Technical Documentation:

Installation in accordance with: Master Data List, PA32-GA8LA, Rev C, dated August 14, 2014.

Operation in accordance with: Flight Manual Supplement, PA32-GA8LA-FM, Rev (-), dated August 14, 2014.

Instructions for Continued Airworthiness: Instructions for Continued Airworthiness including installation, maintenance and service instructions, PA32-GA8LA-105, Rev (-), dated August 14, 2014.

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)" and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -